

# Performance of IR-HARQ-based LDPC Extension Codes in Optical Satellite Systems

NGUYEN Trong Cuong

Computer Communication Lab.,  
The University of Aizu, Japan

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## I. Introduction

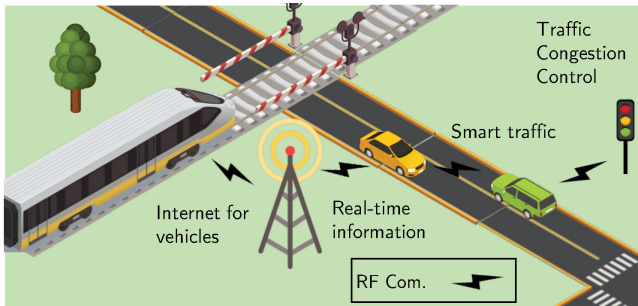
## II. System Description

## III. Results & Discussions

# Internet of Vehicles (IoVs)

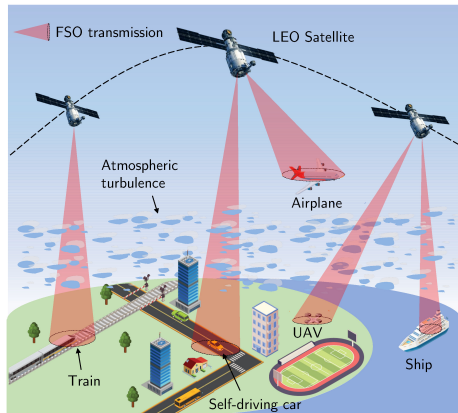
**Internet of Vehicles (IoVs):** The network of vehicles and related entities

- The current network supporting IoV may face **two limitations**
1. Radio-frequency (RF) spectrum scarcity  $\implies$  **Restricted data rate (Mbps or lower)**
  2. Limited coverage of terrestrial infrastructure  $\implies$  **Cannot enable ubiquitous connections**



# Optical Satellite-Assisted IoV

1. **Data rate solution:** Free-space Optical (FSO) Commun.
  - Infrared wavelength (700-1600 nm)
  - Higher data rate ( $\sim$  Gbps or even Tbps) thanks to a vast range of unlicensed bandwidth
2. **Coverage solution:** Low-earth orbits (LEO) Satellites
  - Altitude:  $\leq 2000$ km
  - Low latency compared to other types of satellites
  - Global coverage



➔ Optical LEO-satellite communication is promised to become an enabling technology for the future applications of the IoVs.

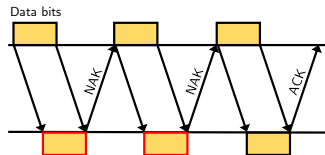


# Possible Solutions: Reliable Trans. Protocols

## Possible solutions: Automatic Repeat Request (ARQ), Error Correction Code (ECC), and Hybrid ARQ

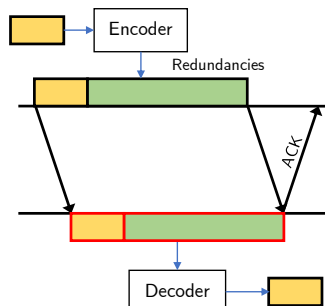
### 1. Automatic Repeat Request (ARQ):

- Retransmit frame when it is erroneous
- May not be efficient as it requires many retransmissions when the channel is deep fading  $\implies$  High latency



### 2. Error Correction Code (ECC):

- Add redundancy to correct a number of errors at the receiver
- May not be efficient as the redundancy is always transmitted  $\implies$  Lower throughput when the channel condition is good



# Possible Solutions: Reliable Trans. Protocols (Cont.)

## 3. Hybrid ARQ: Combination of ARQ and ECC

### Type-I (TI)-HARQ:

- Transmit the same encoded frame every round


### Chase Combining (CC)-HARQ:

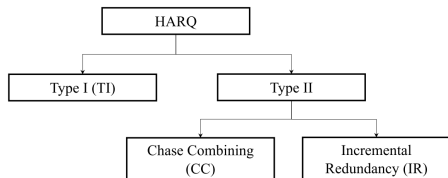
- Combining a number of encoded frames in a single frame based on a maximum likelihood criterion

**Disadvantages:** TI- and CC-HARQ transmit encoded frames every round  $\implies$  Lower throughput when the channel condition is good

### Incremental Redundancy (IR)-HARQ:

- Transmits redundancy when data frame gets errors

 **IR-HARQ** is a potential candidate for the reliable transmission issue of the considered system.



# Literature Review

Currently, there have been some studies addressing the HARQ design for optical satellite systems [1-3].

Error-correction codes of these designs focus on **convolutional code** and **Reed-Solomon code**.

## Convolutional code:

- Work on a continuous stream of data bits
- Use the Viterbi algorithm to decode

## Reed-Solomon code:

- Symbol-based error correction code
- Constructed based on the theory of Galois fields

[1] Hung D. Nguyen *et al*, "Throughput and delay performance of cooperative HARQ in satellite-HAP vehicle FSO systems," in *Proc. IEEE Veh. Technol. Conf.*, 2021.

[2] Hoang D. Le *et al*, "On the design of FSO-based satellite systems using incremental redundancy hybrid ARQ protocols with rate adaptation," *IEEE Trans. Veh. Technol.*, 2022.

[3] Hoang D. Le *et al*, "Fso-based space-air-ground integrated vehicular networks: Cooperative harq with rate adaptation," *IEEE Trans. Aerosp. Electron. Syst.*, 2023.




# Literature Review

One property of the considered system is the preference for long data frames

- Effectively make use of high data rate ( $\sim$  Gbps) and long coherent time (in order of milliseconds)

However, the ECCs of the current designs may not be efficient when the data frame is long

- **Convolutional Code:** Long data frames result in exponentially increasing complexity of Viterbi decoder
- **Reed-Solomon Code:** Long Reed-Solomon codes implemented on a large-size Galois field may be infeasible due to the high complexity

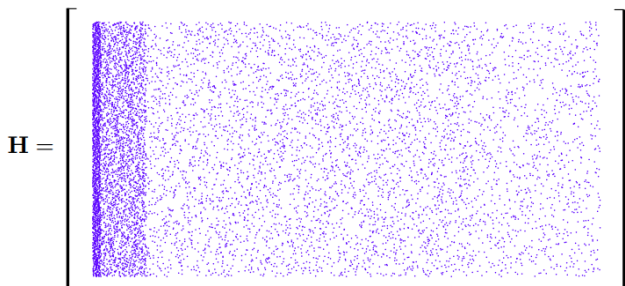
 It is necessary to have a design of proper ECC

# Low-density Parity Check (LDPC) Code

**Low-density Parity Check (LDPC) code**, which is a class of linear block code with sparse parity check matrix, *show great advantages in long frame length regimes.*

- When the frame length increases, the performance of the LDPC code can *approach Shannon's theoretical limit.*
- Thanks to the sparse parity check matrix, it can retain *low decoding complexity when the frame length increases.*

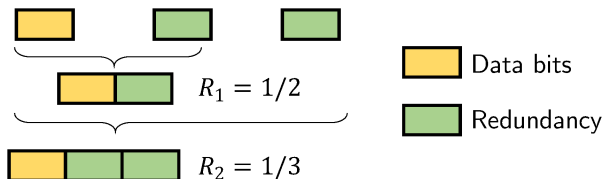
LDPC code is a potential candidate for considered systems.



# Rate-compatible (RC)-LDPC Code Family (1)

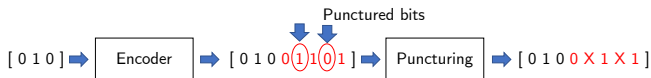
To facilitate the operation of the IR-HARQ, we consider **rate-compatible (RC)-LDPC** code.

- A high code rates are nested in a lower code rate  $\implies$  A low-rate coded frame can be constructed by adding redundancy to a higher-rate coded frame
- Allow the change of code rates with only one pair of encoder/decoder  $\implies$  Reduce complexity



## Rate-compatible (RC)-LDPC Code Family (2)

1. **Puncturing:** Selected bits are removed from an encoded frame to obtain a frame with a higher code rate



**Limitation:** Performance degradation of higher-rate codes

2. **Code extension:** Extend the parity check matrix of a higher-rate code to obtain that of lower-rate codes

$$H_{1/3} = \begin{array}{|c|c|} \hline H_{1/2} & 0 \\ \hline \hline \hline \end{array}$$

# Motivations

- One of the challenging issues in optical satellite systems is the unreliable downlink channel.  
⇒ **IR-HARQ** may offer better performance compared to ARQ and ECC for the desired system.
- Convolutional code and Reed-Solomon codes applied in the current design may not be efficient.  
⇒ **LDPC code**, which has not been considered in the literature, is a potential solution for the design of IR-HARQ in such systems.
- To support the IR-HARQ, a proper design of the RC-LDPC code family is necessary.  
⇒ **The RC-LDPC code family derived by code extension** is a more suitable approach compared to the one by the puncturing method.



The IR-HARQ-based RC-LDPC code extension is a promising candidate for optical satellite systems.

# Goals of the Study

1. We consider a design of the IR-HARQ-based LDPC code extension for optical satellite-aided IoV systems.
2. We study the performance of the design in terms of goodput and energy efficiency.
3. From the obtained results, we discuss the selection of a proper transmitted power

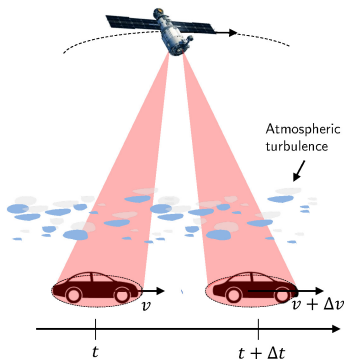
# Outline

I. Introduction

II. System Description

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# System Model



**System model:** Optical downlink channel from an LEO satellite to a ground vehicle

**FSO Channel Model:**

- Turbulence Fading
- Turbulence Attenuation
- Pointing error

**Pointing Error Model:**

- The vibration of the satellite
- The sudden change in vehicle's velocity over a short period of time

**Considered Link-layer Solution:** IR-HARQ based on RC-LDPC Code Extension

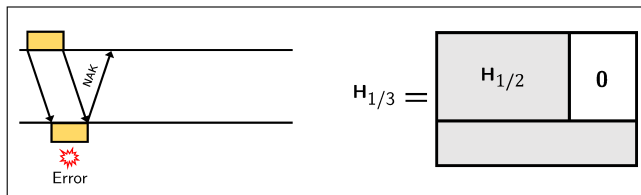


# IR-HARQ based on RC-LDPC Code Extension

## Operation Description

- RC-LDPC code family includes  $N_r$  code rates:  $C_1 > C_2 > \dots > C_{N_r}$
- $N_r$  is also the number of operated rounds of the IR-HARQ
- First round is the transmission of an LDPC-coded frame of rate  $C_1$
- Subsequent round is the transmission of a frame with only redundancy
- This redundancy is combined with the received frame(s) to form a new frame of rate  $C_2 > \dots > C_{N_r}$

**Example:** A systematic RC-LDPC code family  $(1, 1/2, 1/3)$

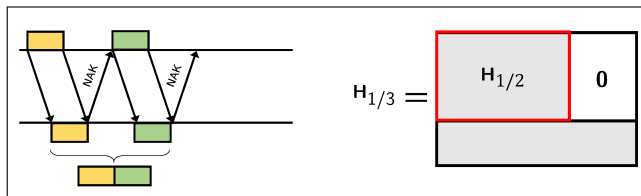


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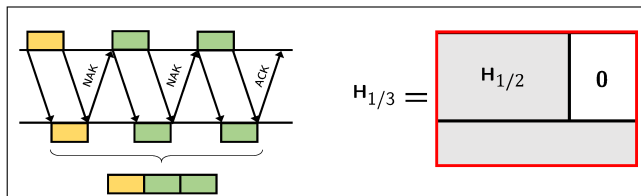


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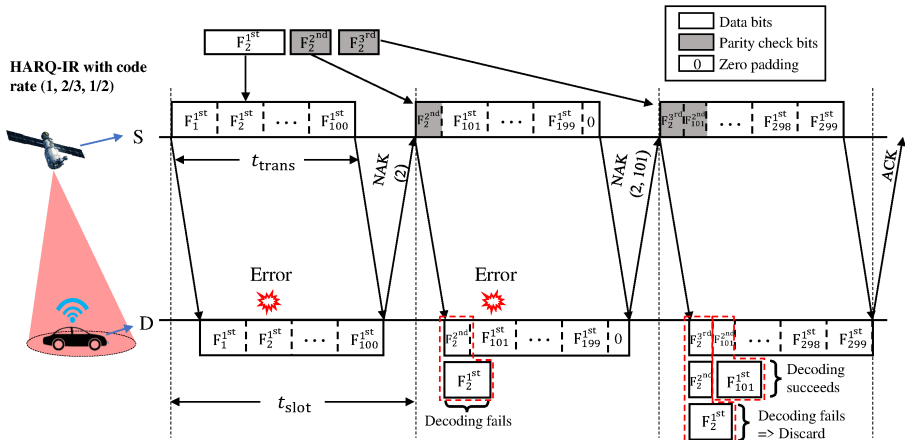
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# An Example of Data Transmission in the Considered System



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# System Parameters

Table: System Parameters

Name	Symbol	Value
<b>LEO Satellite Parameters</b>		
LEO satellite altitude	$H_s$	500 km
Zenith angle	$\xi$	30°
Divergence half-angle	$\theta$	20 $\mu$ rad
Bit rate	$R_b$	1 Gbps
Burst duration	$t_{burst}$	6 ms
Optical wavelength	$\lambda$	1550 nm
Jitter angle	$\theta_{jt}$	2 $\mu$ rad
<b>Vehicle Parameters</b>		
Vehicle altitude	$H_v$	1.5 m
Aperture radius	$r_a$	5 cm
Noise standard deviation	$\sigma_n$	$10^{-7}$ A/Hz
Detector responsivity	$\mathfrak{R}$	0.9
Standard deviation of the velocity variation	$\sigma_v$	4 m/s
<b>Other Parameters</b>		
Atmospheric altitude	$H_a$	20 km
Rms wind speed	$w_{wind}$	21 m/s
Ground turbulence level	$C_n^2(0)$	$10^{-14}$ m <sup>-2/3</sup>

# Performance Evaluation Metrics

**Goodput:** The successfully transmitted data bits per second

$$\text{Goodput} = \frac{\# \text{ of successfully transmitted data bits}}{\text{Simulated time}}$$

**Energy Efficiency:** The successfully transmitted data bits per joule

$$\text{Energy Efficiency} = \frac{\text{Goodput}}{\text{Transmitted power}}$$

# Goodput Comparison among Link-layer Solutions

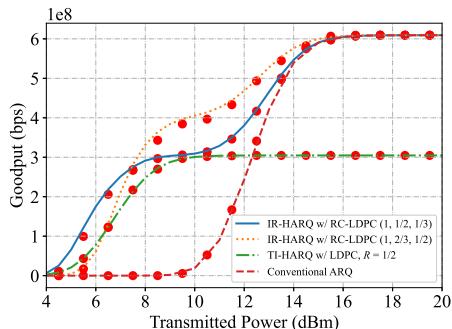


Figure: Goodput versus transmitted powers for different retransmission-based schemes.

- IR-HARQ outperforms TI-HARQ and ARQ in terms of goodput and energy efficiency.
- The trade-off between energy efficiency and goodput

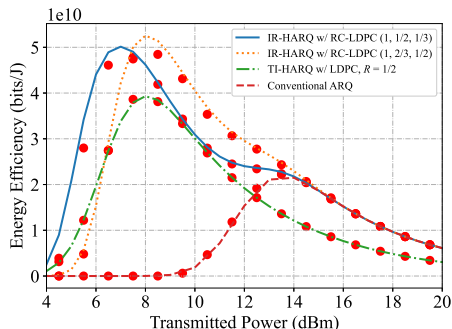
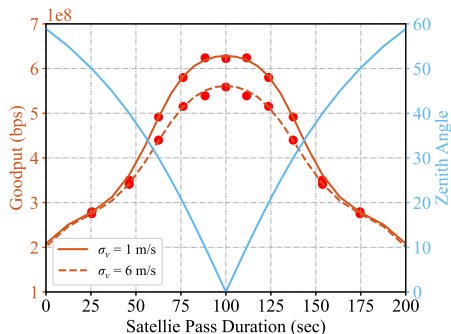


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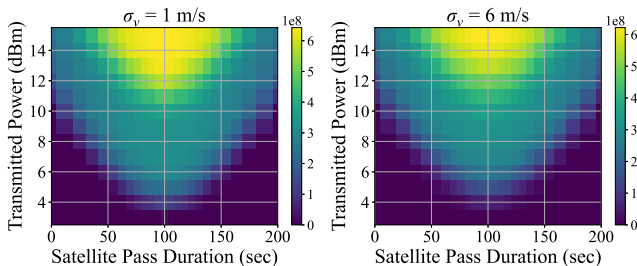
# Goodput Performance over A Satellite Pass Duration



**Figure:** Goodput performance during a satellite pass duration for different values of velocity variation. The transmitted power is 13 dBm.

- **Satellite pass duration:** the period that the satellite may be available for line-of-sight communication.
- The goodput performance changes accordingly with the change of zenith angle.
- The impact of pointing errors is strongest when the beam footprint is smallest, or  $\xi = 0^\circ$ .

# Goodput Performance over A Satellite Pass Duration (cont.)



**Figure:** Goodput performance during a satellite pass duration with different values of transmitted power.

- We can determine the transmitted power to (1) maximize energy efficiency while (2) maintaining a targeted value of goodput during a (3) predefined zenith angle value.
- For example, when  $\sigma_v = 6 \text{ m/s}$ , the transmitted power should be selected as 13 dBm to retain the targeted goodput of 400 Mb/s for the zenith angle  $\xi < 40^\circ$ .

# Conclusion

1. Consider an IR-HARQ-based LDPC code extension design to address the unreliable transmission issue of optical satellite-assisted vehicular networks.
2. Highlight the effectiveness of the IR-HARQ compared to TI-HARQ and ARQ in terms of goodput energy efficiency
3. Discuss the proper selection of transmitted power for the considered system.